

A photograph of a rocket launch, showing the rocket ascending diagonally from the bottom left towards the top right. The rocket has a black body with yellow and white markings. A white plume of smoke or vapor is visible at the base of the rocket. The background is a clear blue sky.

California Launch Vehicle Education Initiative (CALVEIN)

An Overview & Flight Opportunities for “CanSats” and Other Payloads

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CubeSAT Workshop

JPL

April 4-5, 2002

Outline

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California Launch Vehicle Education Initiative

- CALVEIN objectives and summary
- Pilot program: Prospector-1 rocket development
- Development and flight of Prospector-2
- Aerospace System Design curriculum
 - Thrust vector control system
 - Aerospike engine
 - Integration into P-2 and development of P-3
- CALVEIN lessons learned
- Flight opportunities – “*Reusable vehicles are already here*”
- Workshop: “What’s next?”

CALVEIN Objectives and Summary

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- Objectives:
 - Provide engineering students of all levels with hands-on education
 - Provide upper-division engineering students with system development experience: from requirements definition to hardware dev. and flight
 - Stimulate interactions between small payload dev. & launch communities
- Spring 2001: *“From an empty lab to the flight of Prospector-1”*
- CSA/CTTCA funding received in June 2001 to expand the program
- “System integration lab.”: => P-2
 - Get hands-on experience by integrating rocket components
 - Develop/improve reusable rocket subsystems
- Aerospace System Design Curriculum: => P-3
 - From requirement definition to flight tests
 - Develop aerospike engine & thrust vectoring system
- Workshop in June 2002
 - Focused on low cost RLV development
 - Small payloads

Pilot Program: Spring 2001

- Initiated collaboration with Garvey Spacecraft Corp. (GSC)
- Established ***goal of launching before the summer***
- Objectives: provide hands-on experience to engineering students
- GSC provided a “kit rocket” based on their Kimbo V vehicle as well as support for assembly and operations – Prospector-1 (P-1)
- P-1: 160 lb, 10 in dia., 12 ft long, LOX/ethanol, ablative engine, 7,500 ft altitude
- Students (ME & AE) developed a 1000 lbf thrust engine (Kimbo-V had a 500 lbf engine)

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Pilot Program: Spring 2001



Prospector-1: Spring 2001

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**June 3, Flight &
recovery of
Prospector 1 (P-1)**



**May 20,
Static fire
of student
engine...**



System Integration Lab. – P-2

- P-2: “Basic” vehicle with medium-level modifications
- Students of all levels integrating the components provided by GSC



- New subsystem development/improvements:
 - Initiated digital telemetry system dev. (EE)
 - Improved 1000 lbf engine design and manuf. (AE & ME)
 - Filament-wound composite aeroshell from ACPT
 - New student-developed recovery system
- Stanford provided payload and recovery electronics

Telemetry System Development

- Telemetry Project builds upon previous experiments
- Implement real-time digital telemetry, GPS tracking
- Utilize amateur radio equipment, protocols
- Stanford providing technical guidance: based on small satellite technology
- Could become standard for other rocket projects

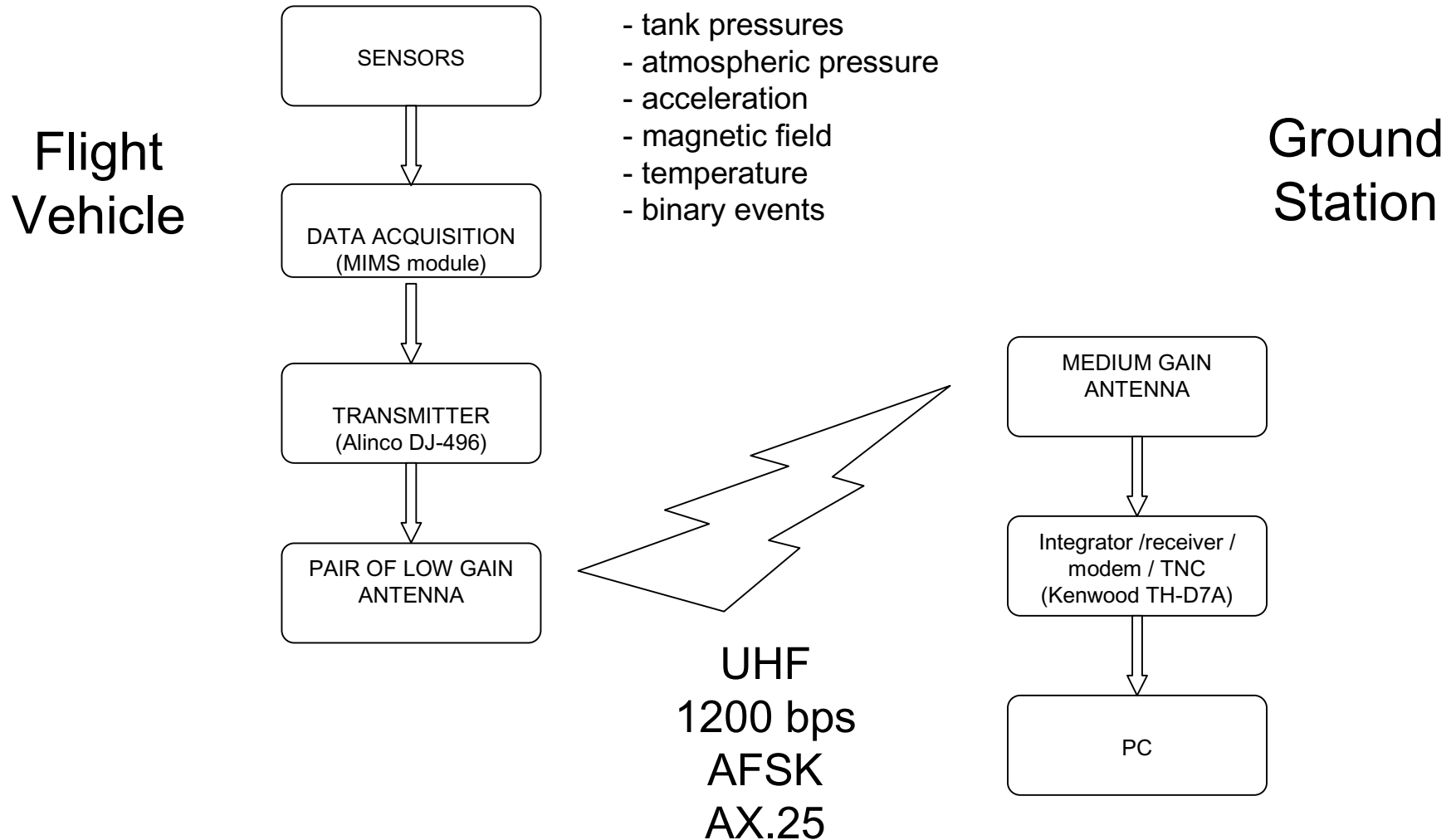


*EE students fabricating
electronics box for Prospector 2*

Telemetry System Development

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Dec. 1, 2001: P-2 Static-fire Test

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- Static fire test of new 1000 lbf engine
- Use of static test stand developed by GSC and integrated into the RRS VTS-2
- Validation of P-2 airframe
- Flight of telemetry system proof-of-concept onboard Kimbo-IX



K-IX with “proof-of-concept” telemetry system

P-2 Launch and Recovery

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- Flight of telemetry system
- Payload deployment system:
Stanford students
- Stanford flight computer: 6000 ft
peak altitude

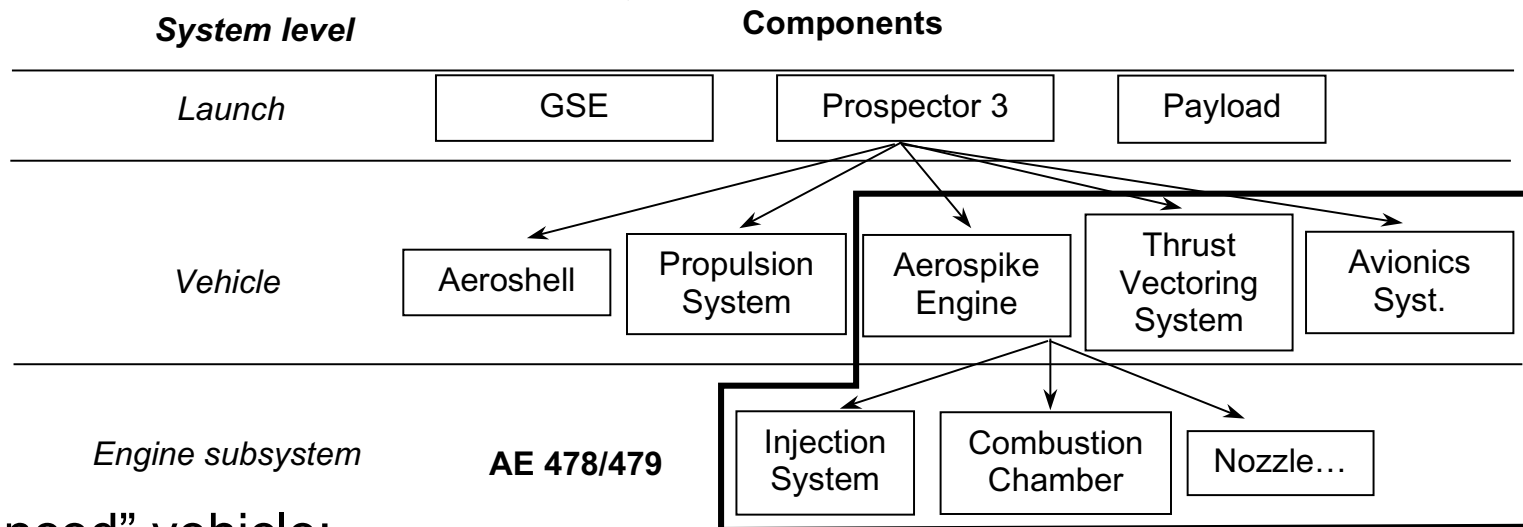


System Design Curriculum

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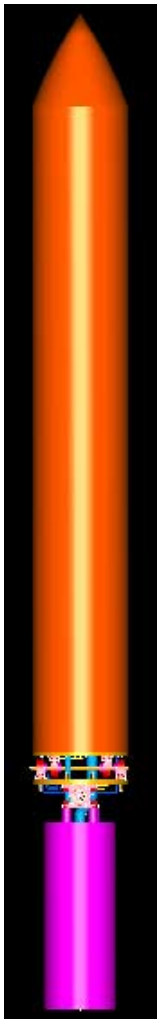
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- Aerospace System Design Curriculum (2 semester design course)
 - Systems Requirement & Functional Review; Oct. 01
 - Preliminary Design Review; Dec. 01
 - Critical Design Review; Feb. 02
 - Test Readiness Review; April 02
 - System Verification Review; May 02

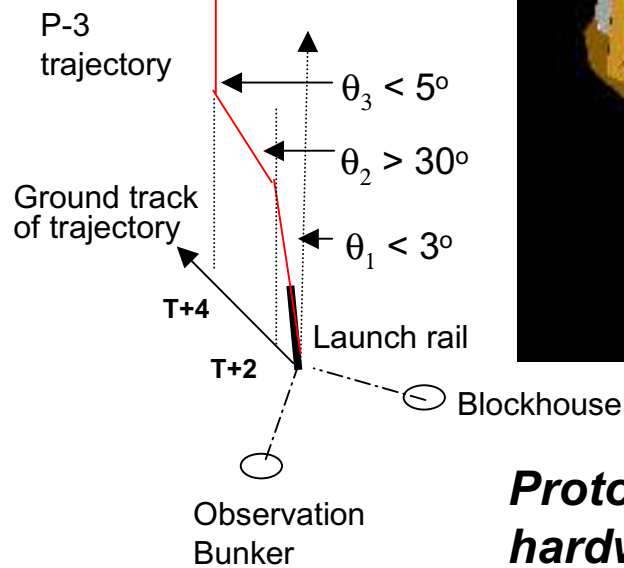


- “Advanced” vehicle:
 - Thrust vector control
 - Aerospike engine

TVC: 2-Axis Gimbal System

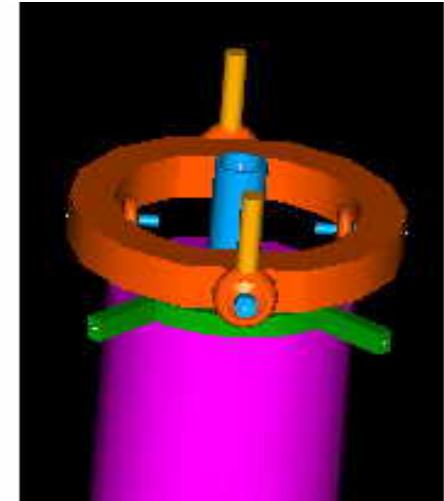
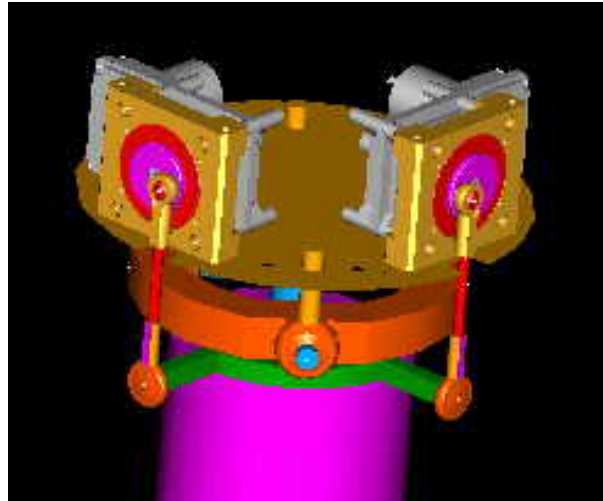


Functional requirements



Prototype hardware

- Static fire: May 2002
- Flight test (P-3) : June 2002

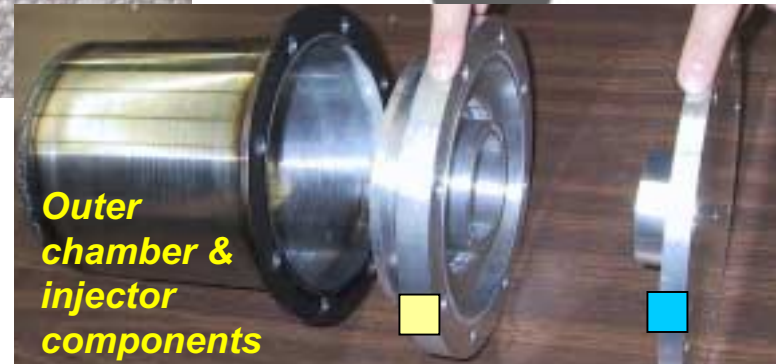
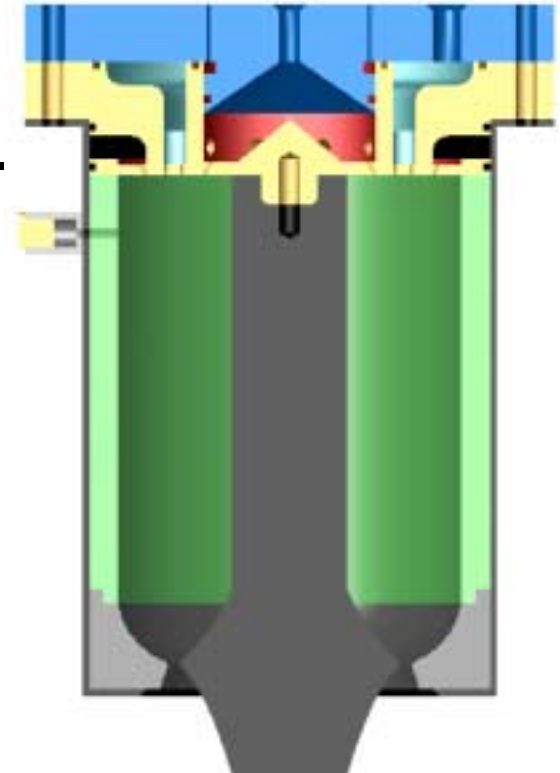
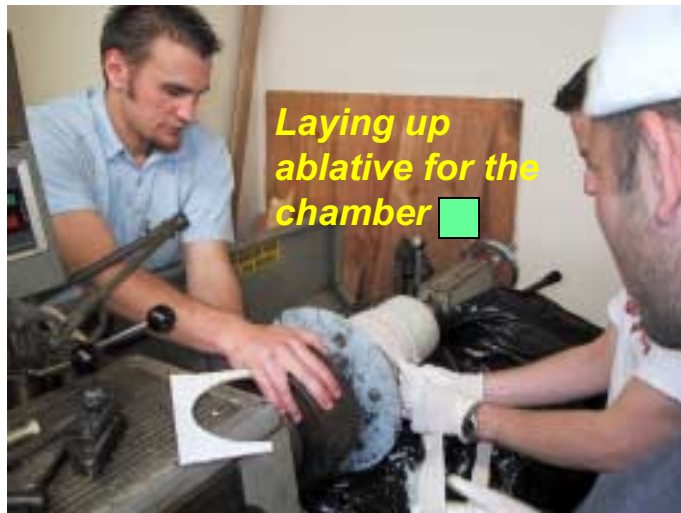


Aerospike Engine

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- Requirements
 - 1000 lbf thrust
 - Min. of 6 sec. Burn
- Static fire: April 20, 02
- Flight test: early May 02
- Reuse P-2 vehicle



Successful flight (of first aerospike engine) will represent a technical milestone in the history of rocket propulsion development

CALVEIN Lessons Learned

- Set achievable goals: technically & in terms of schedule and cost.
6 months – 1 yr. time frame
- Need to establish a schedule with measurable milestones and need to STICK to the schedule
- Students need NOT design/build/test everything
- Providing a “hardware kit” can help drastically jump-start a program
- Focus on incremental improvements
- Student projects need experienced mentors on both technical and management aspects
- New subsystem development should be non-critical path, e.g. engine (P-1 had a 500 lbf backup engine)
- Students put more attention and effort into a piece of hardware which they truly believe is going to fly

Flight Opportunities – Near term

- 10 in diameter x 1 ft length payload bay
- Up to 5 lbs
- Up to 5 g's acceleration



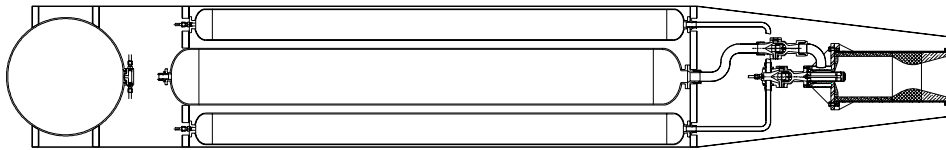
- Altitudes less than 10,000 ft
- No payload deployment
- P-2 with aerospike engine: May 4, 2002 (targeted date)
- P-3 with TVC system : June 2002

Flight Opportunities – Longer term

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- Similar vehicles & flight profiles, possible payload deployment
- Improved vehicles: higher altitudes, larger payloads, etc.:
 - 12,000 lbf. vehicle now in development @ GSC
 - Students already participating



CALVEIN Workshop: “What’s next?”

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- Objectives and scope
 - Coordinate efforts in academia and interests of small aerospace companies regarding:
 - launch vehicles
 - payloads
 - Explore opportunities for adapting small satellite technologies to launch vehicles
 - Address the needs of the small satellite community:
 - “Low” altitude flights
 - Sub-orbital flights
 - The future: achieving low cost orbital flights
- CSULB, mid-June (2002)
- Format:
 - Session 1 (morning): Review of existing programs
 - Session 2 (afternoon): Workgroup: “What’s next?”